Scrial No.: 10/248,967 Confirmation No.: 4437

Applicant: BODIN, Jan-Olof et al.

Atty. Ref.: 07589.117.PCUS00

AMENDMENTS TO THE CLAIMS:

1. (Currently Amended) A liquid fuel rocket engine member (10) comprising: a load bearing wall structure (11, 14) comprising a plurality of cooling channels (11) for handling a coolant flow, wherein the load bearing wall structure (11,14) comprises a curved wall (14), and wherein a wall of each of said cooling channels is attached to said curved wall; and each of the cooling channels (11) having a flow guiding surface (15,16,17,19) extending at an angle to the cooling channel axis for and thereby providing the axial coolant flow with an added radial directional flow component.

2. (Original) The liquid fuel rocket engine member as recited in claim 1, further comprising: the flow guiding surface (15) being incorporated into the channel wall (18).

3. (Original) The liquid fuel rocket engine member as recited in claim 2, further comprising: the flow guiding surface comprising a plurality of grooves in the channel wall (18).

4. (Original) The liquid fuel rocket engine member as recited in claim 2, further comprising: the flow guiding surface (15) comprising a plurality of ribs protruding (15) from the channel wall (18).

5. (Original) The liquid fuel rocket engine member as recited in claim 1, further comprising: the flow guiding surface (16,17,19) comprising a separate structure inside the cooling channel (11).

6. (Original) The liquid fuel rocket engine member as recited in claim 5, further comprising: the structure comprising a helical spiral (19).

7. (Original) The liquid fuel rocket engine member as recited in claim 5, further comprising: the structure having a threaded screw (16, 17).

Serial No.: 10/248,967 Confirmation No.: 4437

Applicant: BODIN, Jan-Olof et al. Atty. Ref.: 07589.117.PCUS00

8. (Currently Amended) A method for manufacturing a liquid fuel rocket engine member (10) having a load bearing wall structure (11, 14) comprising a plurality of cooling channels (11) for handling a coolant flow, said method comprising:

shaping a <u>plurality of sheet metal surface surfaces</u> to provide [a]flow guiding <u>surfaces</u> (15) by <u>surface (15)</u>; folding the <u>plurality of sheet metal surfaces</u> into cooling channels (11); and

forming said wall structure (14) by attaching at least said folded plurality of sheet metals metal surfaces by attaching said folded sheet metals to a wall (14) of an engine member (10) and thereby forming said wall structure.

- 9. (Currently Amended) The method as recited in claim 8, further comprising: wherein said folding of shaping the plurality of sheet metal surface surfaces is performed by stamping grooves into the surface.
- 10. (Currently Amended) The method as recited in claim 8, further comprising: shaping, by stamping, the plurality of sheet metal surfaces surface by stamping to form to have protruding ribs (15) on the surface.